

# BCU-102A-1 Blanking Control Unit Eon P/N 29000-300

## Interface Control Document: Product Data Sheet including O&M Drawing

EON Doc # 29000-350 Rev A – 08/10/21

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P/N:



#### **Product Data Sheet – BCU-102A-1**

(07/25/2021)

Model Number: 1000-300

#### General:

The KC-390 Interference Blanker, (BCU-102A-1), is a customized programmable system utilizing Eon Instrumentation's twenty-year experience in blanking technology. Inputs accept a wide range of voltage levels and pulse frequency. Outputs are generated from a programmable blanking map of inputs to outputs. Bidirectional inputs/outputs are used to accommodate ARINC bus connections. Discrete inputs can dynamically enable or disable individual blanking inputs from the blanking map. The BCU-102A-1 is fully qualified to Environmental and EMI specifications DO-160, 461, 704, and 810.

Please access Eon's website for information and engineering staff for other Eon blankers. Additional Eon product offerings are Video products (splitters/converters/selectors, cameras, monitors, and recorders), Rugged Power Supplies, Audio Systems and Customized System Engineering Development.

#### **BCU-102A-1** Requirement/Feature Description:

Number	Requirement/Feature Description	Embraer Unit
1	The equipment shall accept pulse type signals as inputs. The typical pulse characteristics are: rise time equal or greater than 20 V/ $\mu$ s and the decay time equal or greater than 10 V/ $\mu$ s.	
2	The equipment shall have four (4) single ended inputs available on front panel.	threaded BNC's
3	Internal failure shall not lead to short circuit in inputs and outputs interface.	X
4	The equipment shall have four (4) single ended outputs available on front panel. These outputs shall amplitude of 28V.	threaded BNC's
5	The equipment shall produce output pulses with rise and decay time equal or greater than 30 $V/\mu s$ .	х
6	The equipment shall have four (4) single ended bidirectional input/output available on front panel. These inputs shall accept pulses of 15 ( $\pm 10\%$ ) to 70 V and output with amplitude of 28 V.	
7	The resistive impedance at each input port shall be at least $2 k\Omega$ , when measured relative to ground, shunted by not more than 50 pF capacitance. In order to comply with ARINC 709, Attachment 6.	· · · · · · · · · · · · · · · · · · ·
8	The resistive impedance at each output port shall be at least 20 k $\Omega$ , when measured relative to ground, shunted by not more than 30 pF capacitance. In order to comply with ARINC 709, Attachment 6.	· · · · · · · · · · · · · · · · · · ·

9	Inputs and outputs shall be galvanically isolated by means such as capacitive coupling, inductive coupling, optical coupling or equivalent.	X
10	The equipment shall have four (4) discrete inputs available on front panel.	X
11	Each discrete shall be assigned to a specific input/output interface. When the discrete is high, the corresponding input shall be disabled. When the discrete is low, the corresponding input shall be enabled.	х
12	The discrete signal shall disable the corresponding input/output interface when voltage is between 5 VDC and 30.3 VDC, driving up to 90mA.	x
13	The discrete signal shall enable the corresponding input/output interface when voltage is less than 5VDC.	X
14	The system interfaces shall be robust against inadvertent short circuit.	X
15	Each output shall be capable of combining any combination of inputs.	X
16	Each input signal shall be capable to drive any combination of outputs.	X
17	The equipment shall have configurable blanking matrix.	X
18	The delay time between input signal and the resultant output, when measured between 10% levels of pulses leading edges, shall not be greater than 0.5 µs.	х
19	The delay time between input signal and the resultant output, when measured between 10% levels of pulses trailing edges, shall not be greater than 1 µs.	X
20	The equipment shall be capable of receiving an input signal with max amplitude between 15 and 70 volts.	X
21	The system shall be capable of generating Look Through Blanking Outputs.	X
22	The supplier shall describe the qualification process (software, AEH and problems report) applicable to the system.	Х
23	The equipment shall comply with RTCA-DO-254.	X
24	The Built in Test shall consist of three discrete outputs that are activated when unit is powered on.	X
25	The first discrete shall be activated when 28 VDC power input is available.	X
26	The second discrete shall be activated when power is evident on Pulse Combiner circuit board.	х
27	The third discrete shall be activated when power is evident on Matrix mapping board.	x
28	The system shall operate with 28 VDC aircraft power supply.	Х
29	The Systems shall comply with the electrical bonding and static electricity requirements of the MIL-STD-464A and RTCA/DO-160G;	х
30	Electrical connectors, contact arrangements and associated backshells shall be qualified according with Military Standards in order to achieve easy maintenance and a smaller number of different part numbers.	х
31	It shall be informed for the system what is the expected or designed Maintainability performance (MTTR).	50,000+
32	It shall be informed for the systems what is the expected or designed Reliability performance (MTBF).	50,000+

#### Configuration:

See the attached Outline and Mounting Drawings.

#### Specifications:

Finish: (except for screws, base and connectors): Black Anodize, White Lettering

Input Voltage: 28vdc

Power Consumption: <10.0Watts
Power Dissipation: <2 Watts

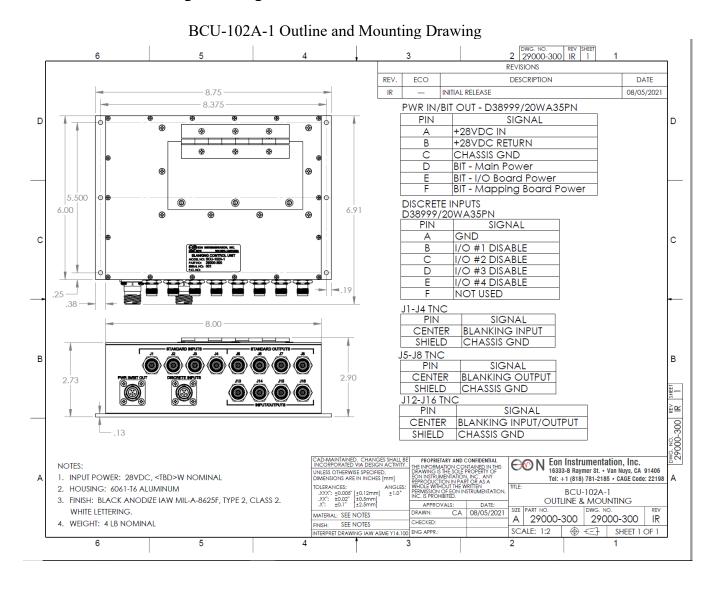
Weight: <3.5 lbs

#### MTBF: 50,000+operational hours

### Qualification (Data by similarity available upon request):

34	The system shall be designed to comply with RTCA-DO-160 according to the following TABLE.	х
	4 - Temp. & Alt.	
	D2 for temp	
	D2 for <40,000 ft	
	A2 for decom and overpressure	X
	5 - Temp. Var. (A)	Х
	6 - Humidity (B)	X
	7 - Shock&Crash (B, aircraft type 2)	X
	8 - Vibration	
	Category R, Curve C&C1	
	Category H, Curve R	X
	9 - Explosion (E or H (if hot spot surfaces))	Х
	10 - Waterproofness (W and Y)	Х
	11 - Fluids (F (spray test only))	Х
	12 - Sand&Dust (S)	Х
	13 - Fungus (F)	X
	14 - Salt (S)	X
	15 - Magnetic Effect (Cat. B)	X
	16 - Power Input (Z)	X
	17 - Voltage spike (Cat A)	X
	18 - Audio Frequency Conducted Susceptibility (MIL-STD-461E Test Category CS101)	x
	19 - Induced Signal Susceptibility (Cat ZC)	Х
	20 - Radio Frequency Susceptibility (Definition according to DAL C, Cat TT)	Х
	MIL-STD-461E Additional EMI/HIRF Req. (MIL-STD-461E Additional EMI/HIRF Req.)	x
	21 - Emission of Radio Frequency Energy (Cat M)	Х
	22 - Lightning Induced Transient Susceptibility (A2G22)	Х
	23 - Lightning Direct Effects (N/A)	Х
	24 - Icing (N/A)	Х
	25 - Eletrostatic Discharge (Cat A)	Х

#### **Outline and Mounting Drawing:**



#### **Connector Specification:**

Conne	ctor Definition		LRU Connector Specification	Mating Connector
Power In/BIT out	38999 series III		TE Connectivity / Deutsch	TE Connectivity / Deutsch
Power In/Bit out		-	D38999/20WA35PN	D38999/26WA35SN
Disercte Innuts	its 38999 series III	10_112	TE Connectivity / Deutsch	TE Connectivity / Deutsch
Discrete Inputs			D38999/20WA35PN	D38999/26WA35SN
Trigger Input	TNC 50 Ohm	J1-J4	Amphenol RF	Amphenol RF
Blanking Output	TNC 50 Ohm	J5-J8	Amphenol RF	Amphenol RF
Input/Outputs	TNC 50 Ohm	J13-J16	Amphenol RF	Amphenol RF